

NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

WASHINGTON, D.C. 20546

REPLY TO
ATTN OF:

October 14, 1970

TO: USI/Scientific & Technical Information Division
Attention: Miss Winnie M. Morgan

FROM: GP/Office of Assistant General
Counsel for Patent Matters

SUBJECT: Announcement of NASA-Owned
U.S. Patents in STAR

In accordance with the procedures contained in the Code GP to Code USI memorandum on this subject, dated June 8, 1970, the attached NASA-owned U.S. patent is being forwarded for abstracting and announcement in NASA STAR.

The following information is provided:

U.S. Patent No. : 3,145,874

Corporate Source : Calif. Institute of Technology

Supplementary
Corporate Source : Jet Propulsion Laboratory

NASA Patent Case No.: XAC-00731

Please note that this patent covers an invention made by an employee of a NASA contractor. Pursuant to Section 305(a) of the National Aeronautics and Space Act, the name of the Administrator of NASA appears on the first page of the patent; however, the name of the actual inventor (author) appears at the heading of Column No. 1 of the Specification, following the words "... with respect to an invention of. . . ."



Gayle Parker

Enclosure:
Copy of Patent

FACILITY FORM 602

N71-15960	
(ACCESSION NUMBER)	(THRU)
4	00
(PAGES)	(CODE)
(NASA CR OR TMX OR AD NUMBER)	(CATEGORY)
	11

COSATI 14B

N71-15960

Aug. 25, 1964

3,145,874

JAMES E. WEBB ADMINISTRATOR OF
THE NATIONAL AERONAUTICS AND
SPACE ADMINISTRATION
MEANS FOR CONTROLLING RUPTURE OF SHOCK TUBE DIAPHRAGMS
Filed Oct. 22, 1962

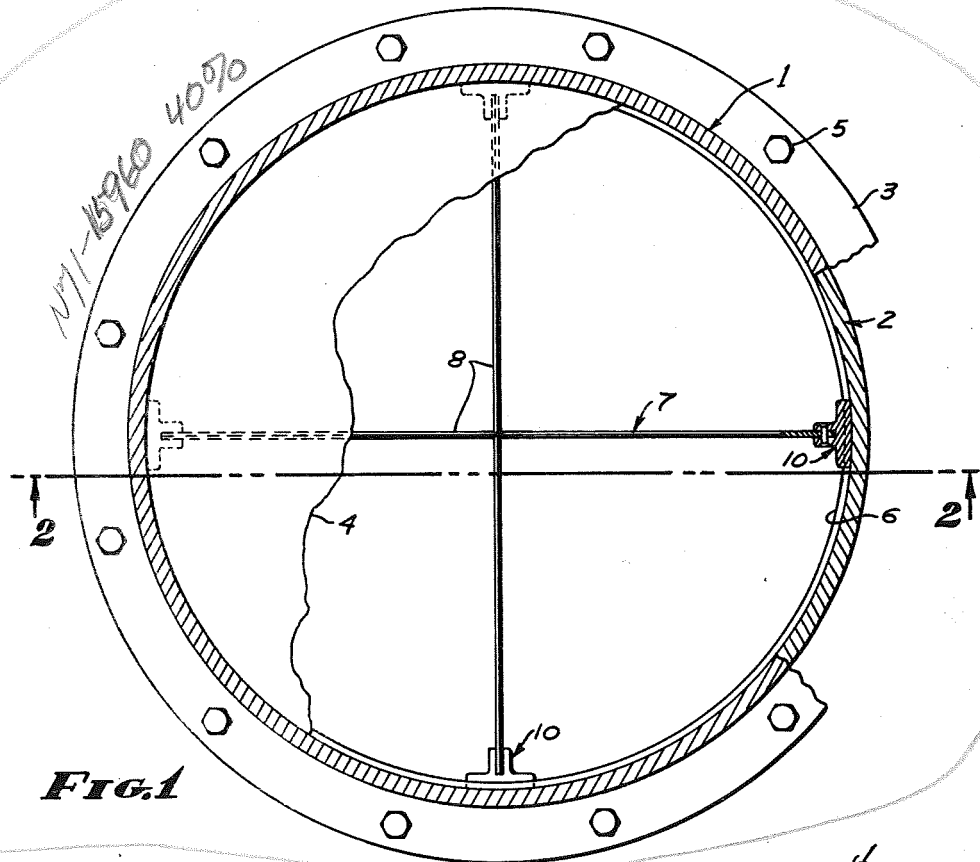


FIG. 1

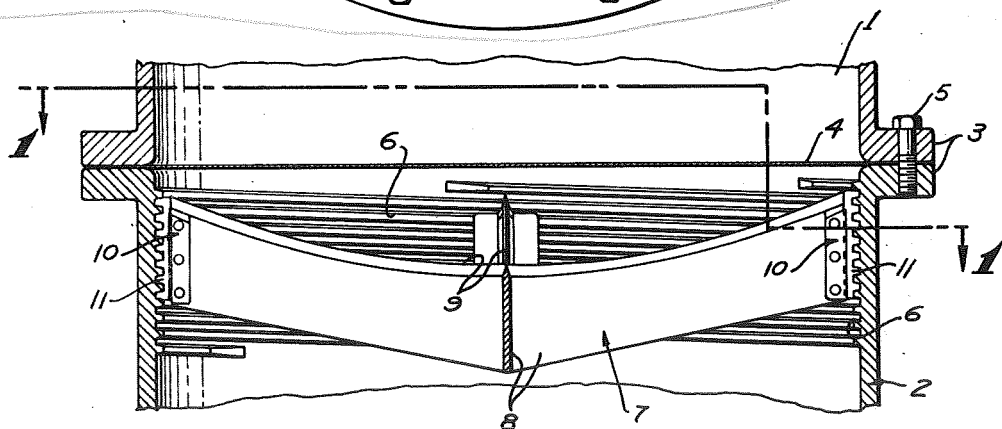


FIG. 2

INVENTOR.
DONALD BAGANOFF

BY *G. W. O'Brien*
Donald C. Kewinney
ATTORNEYS

938

1

3,145,874

MEANS FOR CONTROLLING RUPTURE OF SHOCK TUBE DIAPHRAGMS

James E. Webb, Administrator of the National Aeronautics and Space Administration, with respect to an invention of Donald Baganoff

Filed Oct. 22, 1962, Ser. No. 232,318

1 Claim. (Cl. 220—89)

This invention relates to means for controlling rupture shock tube diaphragms. A shock tube apparatus comprises essentially a compression tube and a low pressure tube separated by a rupturable diaphragm. In operation, the diaphragm is caused to rupture at some predetermined pressure, and the resulting propagation of a shock wave in the low pressure is utilized for various experimental and theoretical studies; such as physio-chemical research, gasdynamic studies, and aerodynamic studies, especially for supersonic research. It is essential that the diaphragm rupture at a predictable pressure.

An object of this invention is to provide a means which ensures precise control of the diaphragm rupture pressure, and which is adjustable to cause rupture of the diaphragm over a wide range of values for diaphragms of a given thickness and material.

A further object is to provide a knife structure disposed under the diaphragm and against which the diaphragm is forced so as to be cut, the position of the knife structure being axially adjustable to precise distances from the diaphragm to control accurately the diaphragm rupture pressure.

A further object is to provide a means for controlling the rupture of a shock tube diaphragm which ensures leafing of the diaphragm; that is, the diaphragm on rupture forms segments or leaves which remain attached and are not blown downstream to interfere with the test or research results.

A further object is to provide a means for controlling rupture of a shock tube diaphragm which permits use of a diaphragm formed of sheet stock without scribe lines or other prepared weakened lines or zones.

With the above and other objects in view, as may appear hereinafter, reference is directed to the accompanying drawings in which:

FIGURE 1 is a transverse, sectional view of a shock tube, showing the means for controlling rupture of the diaphragm and indicating the diaphragm itself fragmentarily, the section being taken substantially through 1—1 of FIGURE 2;

FIGURE 2 is a fragmentary, longitudinal, sectional view thereof, taken through 2—2 of FIGURE 1.

A shock tube comprises at least two sections, namely, an upstream tube section 1 and a downstream tube section 2. The two tube sections are provided with mating flanges 3 between which is clamped a diaphragm 4. The flanges 3 are secured together by bolts 5 or by more elaborate means, not shown, which permits quick coupling and decoupling of the section.

The downstream tube 2 is provided with internal screw threads 6 adjacent its flange 3. Inasmuch as the shock tube may be of relatively large diameter, the screw threads 6 may be of relatively large pitch.

Mounted within the downstream tube 2 is a knife structure 7 which includes a plurality of cutting blades 8 radi-

2

ating from a common center. In the construction illustrated, four such blades are shown, disposed at right angles to each other. The number of blades, however, depends upon the diameter of the shock tube, the blades increasing in number as the diameter is increased. In other words, as few as three blades may be used, or six or eight blades may be used.

Each of the blades 8 is provided with a sharpened upper edge 9 which preferably curves upwardly from the center of the shock tube, that is, the blades 8 define a concave figure.

Each blade 8 is produced at its laterally outer extremity with an end fitting 10, which may be riveted or welded thereto and which includes at its laterally outer side a screw-threaded segment 11. The screw segments 11 of the end fittings 10 engage the internal screw threads 6 so that the knife structure may be adjusted axially on the downstream tube section 2.

Operation of the means for controlling rupture of shock tube diaphragms is as follows:

A diaphragm is used which is capable of stretching or ballooning substantially before rupture. Sheet aluminum or aluminum alloy serves this purpose. However, other sheet metal having lesser or greater strength may be used, depending upon the pressures employed in the upstream shock tube.

The knife structure 7 is located in such a position that the diaphragm will stretch downward into contact with the knife structure before it has been stretched beyond its ultimate strain. Further pressure causes the knife structure 7 to establish lateral lines of concentrated stress in the diaphragm until ultimately the diaphragm ruptures. On rupturing, the diaphragm forms segmental leaves which remain attached to the periphery of the diaphragm and fold downwardly against the walls of the downstream tube section 2.

Prior to the use of the shock tube for study or research purposes, the knife structure 7 is located at various predetermined positions in the downstream tube section 2, and a series of diaphragms formed of a given metal or alloy and thickness is caused to rupture. Careful record is made of the rupture pressure in each case, and the values thus established are used to calibrate the position of the knife structure 7 for subsequent tests. Such calibration is made useful by reason that the rupture pressure for a given location of the knife structure 7 and a diaphragm of given metal or alloy and thickness is remarkably consistent.

While a particular embodiment of this invention has been shown and described, it is not intended to limit the same to the exact details of the construction set forth, and it embraces such changes, modifications, and equivalents of the parts and their formation and arrangement as come within the purview of the appended claim.

What is claimed is:

A shock tube structure, comprising:

- a high pressure tube section;
- a low pressure tube section;
- a diaphragm clamped between said tube sections, said diaphragm being formed of sheet material capable of stretching, thereby to expand axially into said low pressure tube section on application of pressure in said high pressure tube section;
- internal screw-thread means extending circumferentially around and on the inner surface of said

4

UNITED STATES PATENTS

1,415,216	Bingay	May 9, 1922
2,095,828	Nerad	Oct. 12, 1937
2,225,220	Huff	Dec. 17, 1940
2,291,360	Unger	July 28, 1942
2,304,417	Mason	Dec. 8, 1942
2,320,211	Bloom et al.	May 25, 1943
2,788,794	Holinger	Apr. 16, 1957

488,953 Great Britain ----- July 18, 1938